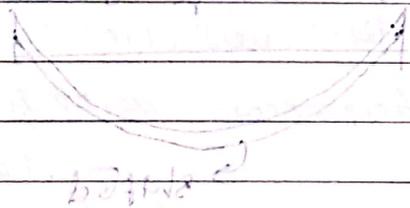
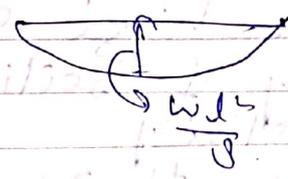
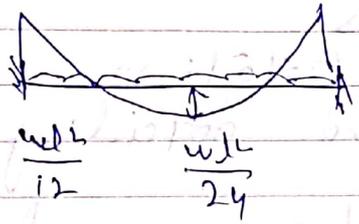


Course:

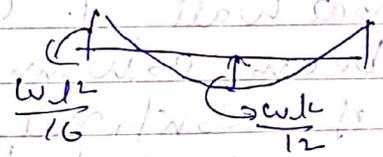
- * Limit state Design philosophy.
- * Redistribution of moments in continuous span Beams.
- * plastic hinge - Concept, rotation Capacity of sections and detailing for ductility.
- * yield line theory for slabs - equilibrium and virtual work methods; shrinkage + creep analysis for stresses in Compression + flexural members; deflection; Shear wall + Coupled Shear wall; beam column joints; prestressing of continuous beams + portal frames; partial prestressing + circular prestressing.



x WSP (f.o.s) + LSD collapse, deflection & cracking partial fact. of s. in material & load. + USP non linear Behavior of sh.

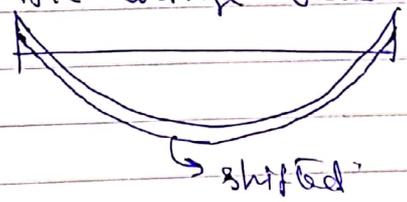


under reinforced → steel will yield & rotation will take place causing redistribution of moment



$$\frac{wl}{12} - \frac{wl}{16} = \frac{wl}{48} = \frac{1}{3} \left(\frac{wl}{12} \right)$$

initial stresses are less in concrete in under reinforced beams & increase after steel yields & provide rotation. moment diagram shift. plastic hinge developed.



Depth → Governed by deflection Criteria

yield line pattern → depend on end condition.



References:

- 1) A. K. Jain → Limit state Design of Con.
- 2) Krishna J + Jain o.p. plain + reinforced Con. vol II
- 3) N. Krishna Raju.

CE 562	Tuesday	wed.	th
	10-11	12-01	11-12

1.5 : 456, 13920, 1243 (prestress)

Limit states

- 1) deflection
- 2) cracking
- 3) collapse → Bending, shear, axial, Torsion
Combined, overloading

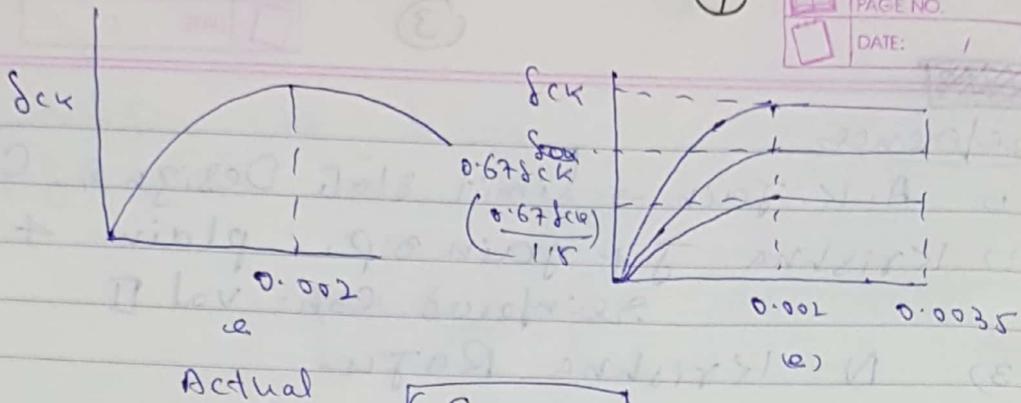
Con. (High st)	<u>standard</u>	ordinary
M-60		M-10
	M-25	M-15
		M-20
M-80		

Steel 1

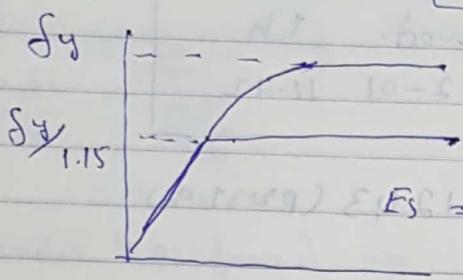
- yield st. (mild steel, well defined yield)
- 0.2% proof stress → corresponding to strain $\epsilon = 0.002$
- IS 875 - loads
- IS 1093 - seismic loads

partial load factors

Combination	D.L	L.L	WL	replace WL → E.L
DL+LL	1.5	1.5	-	
DL+WL	1.5	-	1.5	
Stability DL+WL*	0.9	-	1.5	
DL+LL+WL	1.2	1.2	1.2	

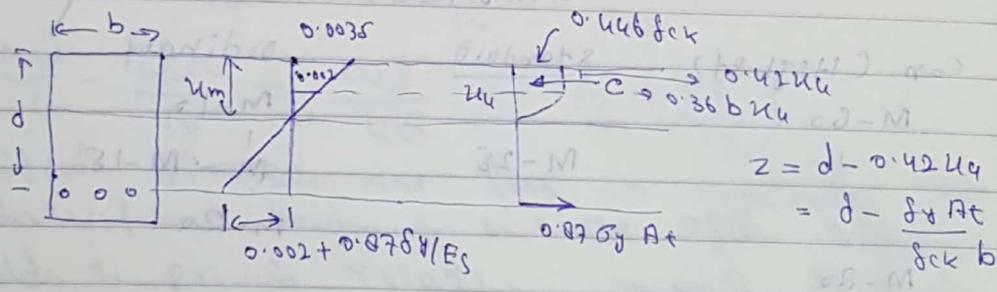


Concrete Idealized



Steel

21/01/20



Assumptions:

- 1) plane section normal to the axis remain plane after bending
- 2) max. strain in concrete at failure is 0.0035
- 3) max. stress in steel $0.67\delta_{ck} / 1.15 = 0.446\delta_{ck}$
- 4) tensile strength of concrete is ignored

1) Balanced Section

Comp. strain in concrete + tension strain in steel reach max. value simultaneously. failure could be sudden + brittle.

→ if steel is less → it will start yielding, con. strain will be less than 0.0035 + steel strain will increase (under-reinforced section) and also con. strain will increase also so that there is some deflection. failure will take place when max. con. strain is reached. + is known as tension failure.

→ over-reinforced sections: con. reaches to max. strain. it is compression failure. The failure is brittle but not as sudden as in case of balanced sections.

when crack takes place, σ value decreases + hence deflection increase. so minimum area of steel is provided.

$$\text{Min. steel} = A_o = 0.07 \frac{bd}{f_y}$$

$$\text{Max. steel} = 4\%$$

$$\text{max. value of } \mu_u = \mu_m$$

$$\frac{\mu_m}{d - \mu_m} = \frac{0.0035}{0.002 + 0.07 f_y / E_s}$$

from code:

f_y	$\mu_m \text{ max } / d$
250	0.53
415	0.48
500	0.46

$$\frac{\mu_m}{d} = \frac{0.0035}{0.0025 + 0.07 f_y / E_s}$$

$$\begin{aligned} M_{u, \text{cn}} &= 0.36 f_{ck} b \mu_m^2 \\ &= 0.36 f_{ck} b \mu_m (d - 0.42 \mu_m) \end{aligned}$$

$$M_{u, \text{steel}} = 0.07 f_y A_{st} (d - 0.42 \mu_m)$$

for design:

$$M_u = 0.07 \sigma_y A_{st} / 0.36 \sigma_{ck} b$$

if $M_u > M_{u,max}$ (M_m)

section is over-reinforced

∴ deflection is governed by con.

$$M_u = 0.36 M_{u,max} \left(1 - 0.42 \frac{M_{u,max}}{\rho} \right) b d^2 \sigma_{ck}$$

if $M_u < M_{u,max}$

then $M_u = 0.07 \sigma_y A_{st} d \left(1 - \frac{\sigma_y A_{st}}{b d \sigma_{ck}} \right)$

Limiting % of steel

$$p_t \text{ lim} = \frac{A_t \text{ lim} \times 100}{b d}$$

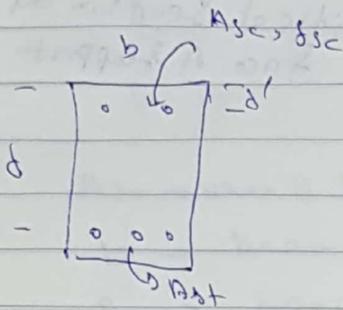
$$A_t \text{ lim} = \frac{0.36 \sigma_{ck} \cdot b \cdot M_{u,max}}{0.07 \sigma_y}$$

$$= \frac{0.36 \sigma_{ck} \left(\frac{M_m}{\rho} \right) \times 100}{0.07 \sigma_y}$$

σ_{ck}	$\sigma_y \rightarrow$			
	250	514	505	
15	1.32		0.76	
20	1.76	0.96	0.94	
25			1.13	
30	2.64			
!				

Doubly Reinforced Section.

$M_F > M_{ulim}$
(factored moment)



$$M_F - M_{ulim} = f_{sc} \cdot A_{sc} (d - d')$$

$$f_{sc} = 0.0035 (M_{max} - d') / M_{max} \cdot E_s$$

Total tension reinforcement $A_{st} = A_{st1} + A_{st2}$

$$A_{st1} = A_{stlim}$$

$$A_{st2} = \frac{A_{sc} f_{sc}}{0.07 f_y}$$

gives for balanced section.

over-reinforced \rightarrow increase tension steel

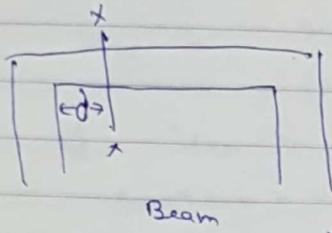
under-reinforced \rightarrow increase compression steel

f_{sc} (stress in comp. steel)

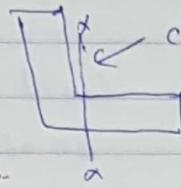
f_y	d'/d			
	0.05	0.10	0.15	0.20
250	217	217	217	217
415	355	353	342	329
500	424	412	395	370

~~Limit state~~

Limit state Design for Shear & Torsion



Beam



critical section at face of support

Nominal shear stress

$$\tau_v = \frac{V_u}{bd}$$

$$\tau_v \neq \tau_{c \max}$$

Con. grade	15	20	25	30	35	40	above
	2.5	2.0	3.1	3.5	3.7	4.0	

Slabs $\rightarrow \tau_v \leq \tau_{ck}$

$\tau_c \rightarrow$ Design shear stress from Table 19.

$\cdot k^2$ for slabs.

over all depth	300	275	250	225	200	175	150
k	1.05	1.1	1.15	1.2	1.25	1.3	

Usually shear reinforcement is not provided in slabs. but if needed

$$\tau_v \neq \frac{1}{2} \tau_{c \max}$$

for slabs otherwise reverse the section.

$$\tau_c = \frac{0.85}{\beta} \sqrt{0.0008 \sigma_{ck} k (\sqrt{1+5\beta} - 1)}$$

$$\beta = \frac{0.0008 \sigma_{ck}}{6.89 p_t} \neq 1$$

$$p_t = \frac{100 A_e}{b_w d}$$

if $\tau_v < \tau_c \rightarrow$ provide nominal reinforcement
as given:

$$\frac{A_{sv}}{b \cdot s_v} \geq \frac{0.4}{0.87 f_y}$$

$A_{sv} \rightarrow$ area of shear stirrup in cross section

$b \rightarrow$ beam width

$s_v \rightarrow$ spacing

$f_y \rightarrow$ yield stress of shear reinforcement $\rightarrow 415$

ii) $\tau_v < \frac{1}{2} \tau_c$ ^{structural}
or member of minor ~~flexural~~ ^{importance} lintel etc.
then we do not provide shear reinforcement.

if $\tau_v > \tau_c \rightarrow$ provide design shear reinf.

$$V_u - \tau_c b d = V_{us} \text{ (shear force to be carried by shear reinforcement)}$$

$$V_{us} = \frac{0.87 f_y A_{sv} \cdot d}{s_v} \text{ (for vert. stirrups)}$$

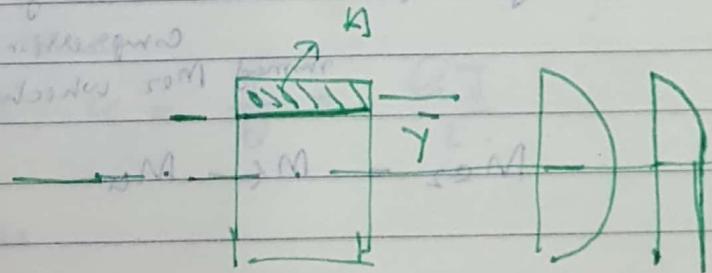
$$= \frac{0.87 f_y A_{sv} \cdot d}{s_v} \times (\sin \alpha + \cos \alpha) \text{ for inclined stirrups}$$

$$\alpha \neq 45^\circ$$

calculate spacing ' s_v ' from above expression.

if $\tau_v > \tau_{c \text{ max}} \rightarrow$ increase section size.

$$\frac{V_u A_y}{I b}$$



Torsion \rightarrow curved beams, bal crney,
water tanks, curved bridges
(Torsion + shear are considered together)

Equivalent Shear

For Torsion critical section will be 'd' from
face of support.

Torsion is max at zero B.M.

$$T_u = \text{Torsional moment}$$

$$V_u = \text{Shear}$$

Eqvt. Shear:

$$V_e = V_u + \frac{1.6 T_u}{b}$$

equivalent nominal shear:

$$\tau_{ve} = \frac{V_e}{bd} \neq \tau_{c \text{ max.}}$$

if $\tau_{ve} < \tau_c \rightarrow$ min shear reinforcement is
provided

if $\tau_{ve} > \tau_c$ then long. + Transverse
shear reinf. are needed

Eqvt. long. moment: M_u ; $T_u =$
(M_{e1})

$$M_{e1} = M_u + M_t$$

$$M_t = T_u \left(1 + \frac{D/b}{1.7} \right)$$

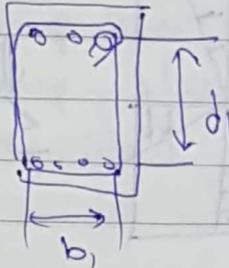
if $M_t > M_u \rightarrow$ long. reinforcement is needed in
compression zone corresponding to
named M_{e2} which is opp. in nature to M_{e1}

$$M_{e2} = M_t - M_u$$

Stirrups → closed hooks

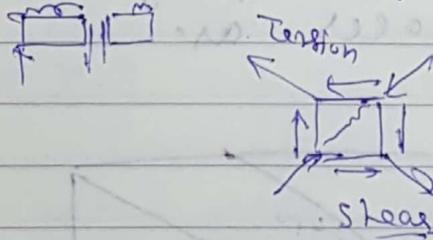
$$A_{sv} = \frac{T_u \cdot S_v}{b \cdot d_1 (0.87 f_y)} + \frac{V_u \cdot S_v}{2.5 d_1 (0.87 f_y)}$$

but not less than $\frac{(T_{ve} - T_c) b \cdot S_v}{0.87 f_y}$



04/00/05

Shear & Torsion

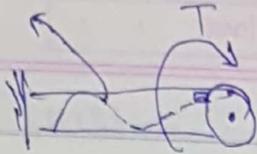


Torsion is not designed alone it is taken in combination with shear & flexure in concrete.

Torsional stiffness, $K = \frac{T}{\theta} = \frac{QT}{L}$

$Q \rightarrow$ shear modulus

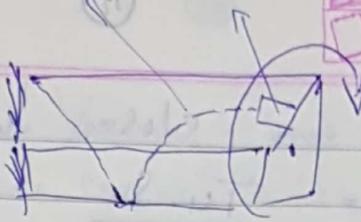
Diagonal crack



$$J = \frac{\pi D^4}{32}$$

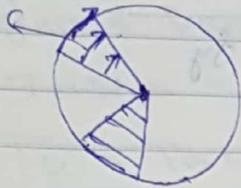
Diagonal crack

Piece



$$J = \frac{1}{3} b^3 y$$

Shear stress

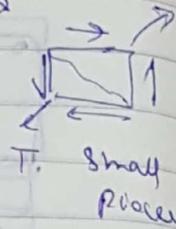
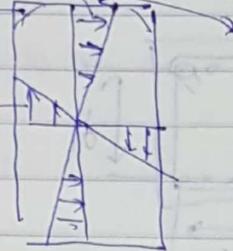


Circular section

not torsional stress in extreme corners.

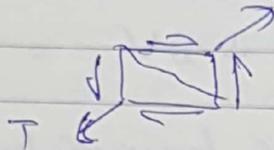
max

Shear stress



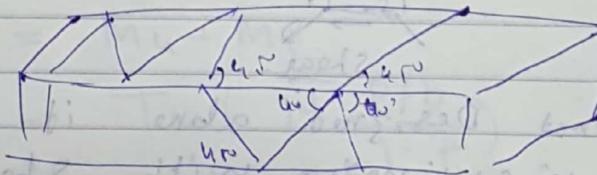
Small piece

Rectangular section



Small ~~circular~~ element taken from circular shaft.

Torsion → has to be stitched by both longitudinal & vertical reinforcement. as the cracks are progressing at 45° over all surfaces. as:



eq. shear

$$V_e = V_u + \left(\frac{1.6}{b} \right) \frac{T_u}{b}$$

equivalent shear

Similarly:

$$M_{ed} = M_u + M_t$$

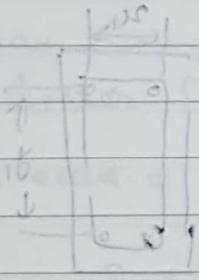
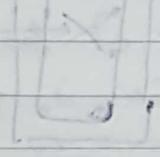
eq. moment.

$$M_t = T_u \left(\frac{1 + D/b}{1.7} \right)$$

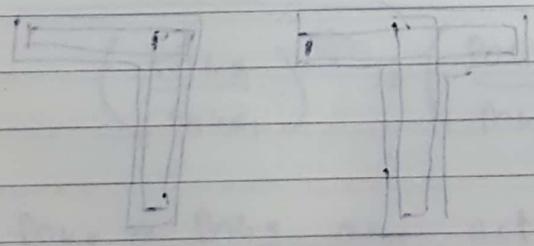
if $M_t > M_u$ → so we have moment $M_{e2} = M_t - M_u$

if then we have to provide longitudinal reinfo-
-ment at the top because M_{e2} creates
Tension.

if $M_t < M_u$ then only comp is actual
comp zone which is taken care by
comp steel provided or min comp steel
provided. M_{e2} is also -ve moment.



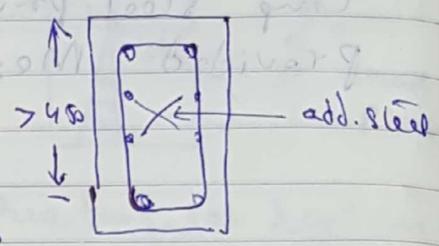
at the top of the beam reinforcement should be provided in the flange and in the web. In case of a beam reinforcement case of the web should be provided in the flange and in the web.



Detailing Provisions

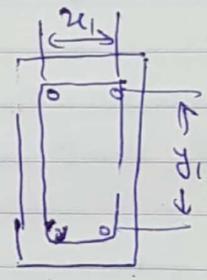
Torsion!

1. long. reinforcement should be placed as near the corners as possible with at least one long. bar in each corner
2. if depth exceeds 450 mm; additional ^{long} bars ^{equally} must be provided on the vertical faces ^{area} of c/s not less than 0.1% of cross section of beam, not greater than width of beam or 300 mm whichever is less.



3. Transverse reinforcement consists of closed rect. hoops placed vertically

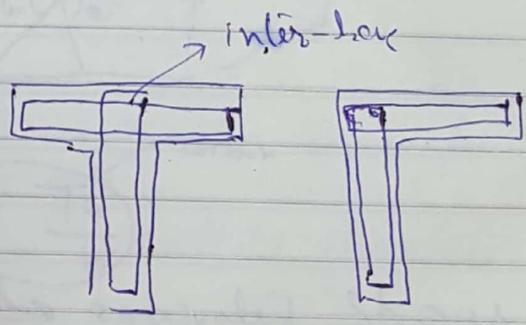
spacing 'Sv' < $2x_1$
 < $(2x_1 + y_1) / 4$
 < 300



T & L Beams

if main reinforcement of the slab is parallel to the LT beam; transverse reinforcement should be provided in the flange equal to 60% of the area of c/s of main reinforcement

The shear reinforcement cages of the keys should interlock



if in T & L Beams if flanges are in tension, then ^{long} torsional reinf. must be distributed over the flange ^{width = span/10 or flange width whichever is less} also. if effective flange width exceeds $1/10$ of span then nominal reinforcement must be provided

Column Design:

1) max. strain due to direct compression is limited to 0.002

if comp. is not uniform but there is no tension on any face then

Max. Comp. $\leq 0.0035 - 0.0075$ of strain on lesser comp. face.

$$\text{min eccent.} = \frac{\text{un-supported length of column}}{50} + \frac{\text{lateral dim}}{30}$$

Sub. to min of 20 mm

Short axially loaded columns

$$P_u = 0.4 f_{ck} A_c + 0.67 f_y A_{sc}$$

Short column loaded with axial load and uni-axial moment \rightarrow Design code SP-16

Bi-axial Bending + Comp

$$\left(\frac{M_{ux}}{M_{uxl}} \right)^{\alpha_n} + \left(\frac{M_{uy}}{M_{uy1}} \right)^{\alpha_n} \leq 1.0$$

M_{ux} & M_{uy} are actual moments about X & Y axes with the given compressive load.

M_{ux1} & M_{uy1} are uniaxial moment capacities with the axial load

α_n depends on P_u / P_{u2}

$$P_u = 0.45 f_{ck} A_c + 0.75 f_y A_{sc}$$

$$\frac{P_u}{P_{u2}} = 0.2 \text{ to } 0.8 \rightarrow \alpha_n \text{ varies linearly from } 1 \text{ to } 2$$

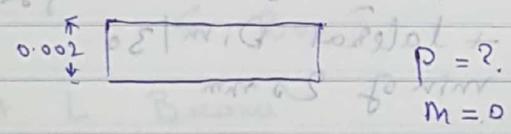
< 0.2 then $\alpha_n = 1$
 > 0.8 then $\alpha_n = 2$

10/8/05

Short Comp. members:

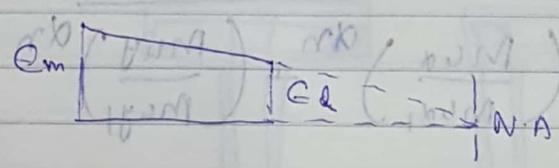
axially loaded $P_u = 0.4 f_{ck} A_c + 0.67 f_y A_{sc}$

for eccentricity up to 0.05 of lateral dimension



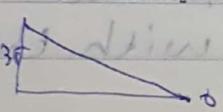
for no eccentricity $e = 0$ N.A \rightarrow infinity

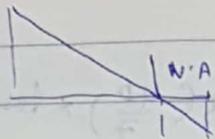
Comp. strain taken axially loaded + flexural
 $= 0.0035 - 0.75$ of comp. strain on load
comp edge.



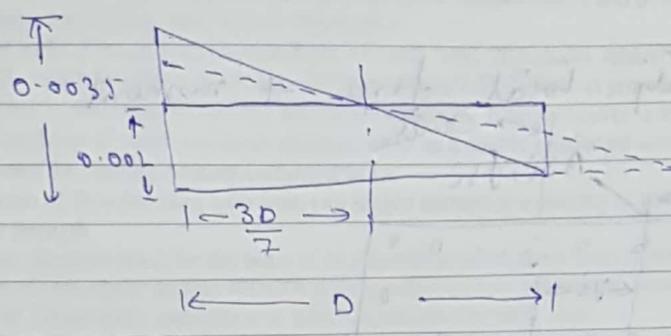
$$\epsilon_m = 0.0035 - 0.75 \epsilon_c$$

For $\epsilon_c = 0 \rightarrow \epsilon_m = 0.0035$

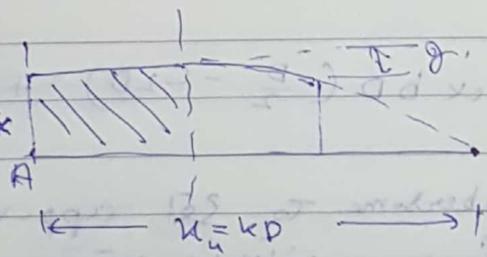
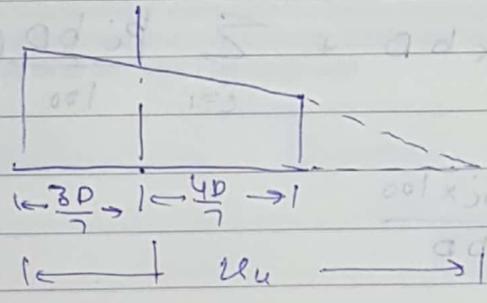




Combine:



General case:



$$\begin{aligned}
 g &= 0.446 \sigma_c k \left[\frac{4/7 D}{kD - \frac{3D}{7}} \right]^2 \\
 &= 0.446 \sigma_c k \left(\frac{4}{7k - 3} \right)^2
 \end{aligned}$$

$$\begin{aligned}
 \text{area of stress block} &= 0.446 \sigma_c \frac{4}{7} D - \frac{2}{3} \left(\frac{4}{7} D \right) \\
 &= C_1 \sigma_c D
 \end{aligned}$$

when multiplied by 'b' gives total compressive force. (see A)

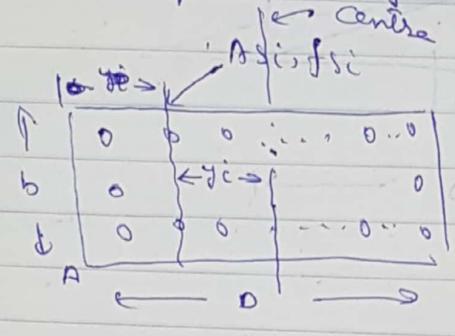
Taking moments about A.

$$= 0.44 b f_{ck} D \left(\frac{D}{2}\right) - \frac{2}{3} \left(\frac{4P}{7}\right) \times \text{lever arm}$$

$$= M \text{ say.}$$

C.G. distance from compressed edge A = $\frac{M}{A}$

for calculating total force + moment



$$P_u = c_1 f_{ck} b D + \sum_{i=1}^n \frac{p_i b D}{100} (\sigma_{sc_i} - f_{ck})$$

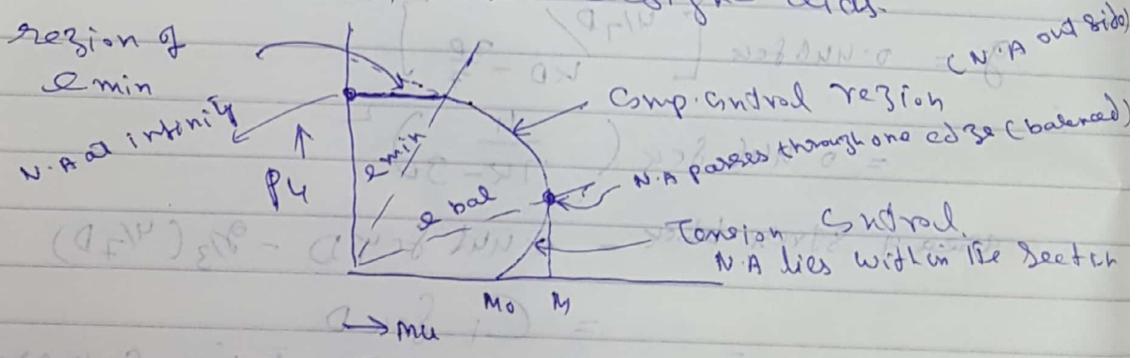
$$p_c = \frac{A_{sc} \times 100}{b D}$$

So moment:

$$M_u = c_1 f_{ck} b D \left(\frac{D}{2} - c_2 D\right) + \sum_{i=1}^n \frac{p_i b D}{100} (\sigma_{sc_i} - f_{ck}) \times y_i$$

it is cumbersome to get app. value, multiply f_{ck} + interior steel

But we usually use Design-aids.



interaction Diagram

Mo

(A)

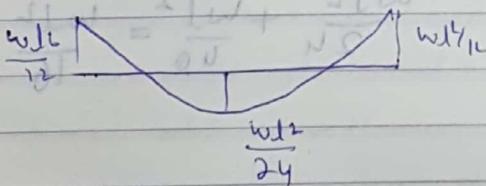
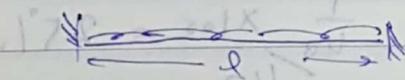
CE-562 Concrete Structures
Tut. 1 Limit State Design

Assignment 2009-10

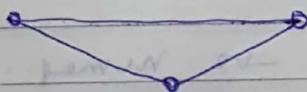
Use M20 Conc. & Fe415 steel. All given forces and moments are factored values.

- 1 (a) Design a beam to resist a factored moment of 60 kNm. Assume $d/b=2$ and $p=0.7\%$. Also calculate the limiting moment and limiting steel area for this section.
- (b) A beam of width 200mm and effective depth 350 with 3nos. 20 mm dia. rebars. Calculate M_u .
- 2 (a) A doubly reinforced beam of width $b=260$ mm and depth $D=520$ mm is provided with 4 nos. 20mm dia. bars in tension and 2 nos. 16 mm dia. bars in compression. Effective cover is 40 mm. Calculate M_u .
- (b) Design the beam of width and depth obtained in 1a as a doubly reinforced section for a moment equal to 1.5 times the limiting moment calculated in 1a.
- 3 (a) The beam of 1b is provided with 8mm two legged stirrups at a spacing of 150mm. Calculate the design shear strength.
- (b) Determine the shear reinf. for the beam of 1b required to resist shear force i) 50kN ii) 100 kN.
- 4 A beam of rectangular section 400×700 deep is subjected to a moment of moment of 100 kNm, a torsion of 15kNm and a shear force of 100kN. Calculate the reqd. reinf.
- 5 (a) Design a short square column to carry a factored axial load of 1500kN assuming 0.8% steel area.
- (b) Redesign the above column to carry a moment of 100 kN in addition to the load. Use design aids.
- (c) Redesign the above to carry the same axial load along with two moments 100kNm and 75 kNm wrt x and y axes (use design aids)
- (d) Design a slender column of cross-section 20×30 effective lengths L_{ex} and $L_{ey} = 4$ m. load = 400 kN and moment in longer and shorter directions are 15kNm and 10kNm. Reinf. is to be distributed equally on all four sides. (use design aids)

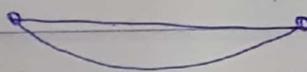
Moment redistribution in Beams



(Elastic B.M.D)



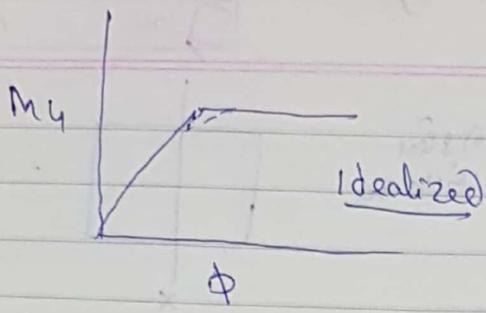
Balanced.



under-reinforced.

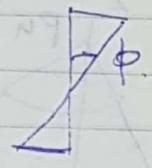
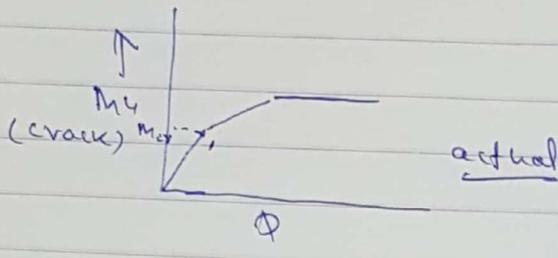
Let us assume that supports have lesser moment capacity. i.e. $wl^2/16$ not $wl^2/12$

in other word if load increases, beam will fail if it is balanced; if it is under-reinforced; hinge will form at support, increase load capacity; then further increase will cause hinge formation at centre, here failure, in a statically determinate structure there is no redistribution capacity

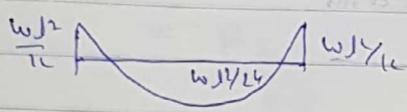


$$f = \frac{M}{I} = \frac{E}{R}$$

$$\phi \propto \frac{1}{R}$$

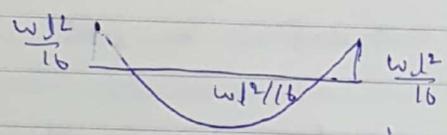


- 1. Code permits up to 30% redistribution of moments in beam.
- 2. In case of a frame of over 4 storeys; if the lateral stability of the structure depends on the frame; then only 10% redistribution of moments is permitted.



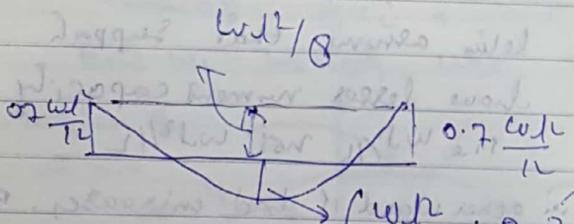
Reduction: $\frac{wL^2}{12} - \frac{wL^2}{10} = \frac{wL^2}{48}$

% = $\frac{1}{48} \times 100 = 25\%$



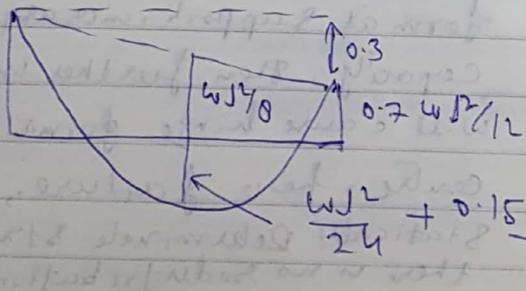
25% reduction

$$\frac{wL^2}{24} + \frac{wL^2}{40} = \frac{wL^2}{16}$$

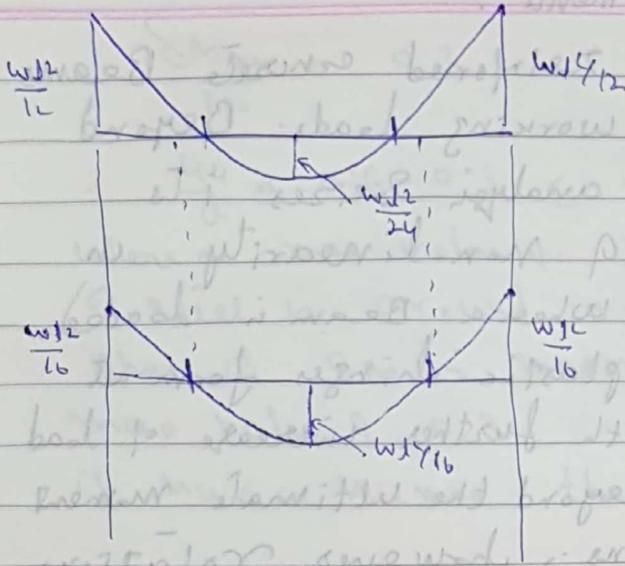


30% reduction

$(\frac{wL^2}{24} + 0.3 \frac{wL^2}{12})$, then the moment becomes more than -ve moment.



$$\frac{wL^2}{24} + 0.15 \frac{wL^2}{12}$$



increasing -ve moment capacity

reduces the demand of -ve moment requirement. not more than 30% is permitted because it will also cause large shifting in point of contraflexure.

point of contraflexure changes due to re-distribution.

When the negative moment capacity is increased, the demand for negative moment is reduced. This is because the beam is able to resist a larger negative moment, so the required negative moment is less. This results in a lower demand curve. The point of contraflexure, where the moment is zero, shifts towards the center of the beam. This is because the beam is able to resist a larger negative moment, so the required negative moment is less. This results in a lower demand curve. The point of contraflexure, where the moment is zero, shifts towards the center of the beam.